06/09/2006

Bank: (Aviation Mechanic Powerplant) Airman Knowledge Test Question Bank

The FAA computer-assisted testing system is supported by a series of supplement publications. These publications, available through several aviation publishers, include the graphics, legends, and maps that are needed to successfully respond to certain test items. Use the following URL to download a complete list of associated supplement books: http://av-info.faa.gov/data/computertesting/supplements.
pdf

1. A03P AMP

If the oil pressure of a cold engine is higher than at normal operating temperatures, the

- A) oil system relief valve should be readjusted.
- B) engine's lubrication system is probably operating normally.
- C) oil dilution system should be turned on immediately.

2. A03P AMP

An engine misses in both the right and left positions of the magneto switch. The quickest method for locating the trouble is to

- A) check for one or more cold cylinders.
- B) perform a compression check.
- C) check each spark plug.

3. A03P AMP

Engine crankshaft runout is usually checked

- 1. during engine overhaul.
- 2. during annual inspection.
- 3. after a 'prop strike' or sudden engine stoppage.
- 4. during 100-hour inspection.
- A) 1, 3, and 4.
- B) 1 and 3.
- C) 1, 2 and 3.

4. A03P AMP

If an engine cylinder is to be removed, at what position in the cylinder should the piston be?

- A) Bottom dead center.
- B) Top dead center.
- C) Halfway between top and bottom dead center.

5. The horsepowe A) shaft horsep B) indicated ho C) brake horse	orsepower.	AMP cating engine is known as the
A) Lower intak B) A backflow	A03P ve overlap promote? e manifold pressure and temperatures. of gases across the cylinder. enging and cooling characteristics.	AMP
constant)? A) A mixture le B) A mixture ri	A03P mixture will result in the highest engine to eaner than a rich best power mixture of .0 ocher than a full rich mixture of .087. eaner than a manual lean mixture of .060.	85.
A) At the end o B) At the end o	A03P Ive overlap occur in the operation of an a of the exhaust stroke and the beginning of the power stroke and the beginning of the compression stroke and the beginn	f the intake stroke. he exhaust stroke.
9. What is the be A) High oil con B) Low compre C) Low oil pres	ession.	AMP
normal operati A) large drop i	on is usually indicated by a n RPM. interruption of both ignition systems.	AMP T or RIGHT during an engine ground check,

11.	A03P	AMP
Engine operating	g flexibility is the abilit	y of the engine to
A) deliver maxim	um horsepower at a	specific altitude.
B) meet exacting	requirements of effic	eiency and low weight per horsepower ratio.
C) run smoothly	and give the desired	performance at all speeds.
12.	A03P	AMP
f the oil pressure		er a wide range from zero to normal operating pressure, the
A) low oil supply		
B) broken or wea	ak pressure relief valv	e spring.
C) air lock in the	scavenge pump intal	ce.
13.	A03P	AMP
A characteristic	of dyna focal engine r	nounts as applied to aircraft reciprocating engines is that the
	-	al flexing of the powerplant.
•		ts at the engine's center of gravity.
,		ine's center of gravity.
,		3 3 3
14.	A03P	AMP
Excessive valve	clearance results in t	ne valves opening
A) late and closir	ng early.	
B) early and clos	sing late.	
C) late and closi	ng late.	
15.	A03P	AMP
		gine have been serviced, in what position should they be
A) Next in firing (order to the one from	which they were removed.
B) Swapped bott		•
C) Next in firing (order to the one from	which they were removed and swapped bottom to top.
4.0	4.00D	AAAD
16.	A03P	AMP
/vnat is tne purp A) To check mao		on a reciprocating engine?
,	satisfactory performa	nce
•	if the fuel/air mixture	
o, to actornine	ii tilo idol/ali illixtule	io adoquato.

17.	A01P	AMP		
Which statement is tru	ue regarding bearings used in high po	owered reciprocating aircraft engines?		
A) The outer race of a radius of the balls.	A) The outer race of a single row, self aligning ball bearing will always have a radius equal to the radius of the balls.			
B) There is less rolling	g friction when ball bearings are used	than when roller bearings are employed.		
C) Crankshaft bearing without overheating.	gs are generally of the ball-type due to	their ability to withstand extreme loads		
18.	A01P	AMP		
The five events of a fo	our stroke cycle engine in the order of	f their occurrence are		
A) intake, ignition, cor	mpression, power, exhaust.			
•	pression, ignition, exhaust.			
C) intake, compression	on, ignition, power, exhaust.			
19.	A01P	AMP		
Which of the following	g is a characteristic of a thrust bearing	used in most radial engines?		
A) Tapered roller.		Ç		
B) Double row ball.				
C) Deep groove ball.				
20.	A01P	AMP		
What is the principal advantage of using propeller reduction gears?				
		accompanying increase in engine RPM.		
,	ine RPM to be increased with an acco n at a lower, more efficient RPM.	ompanying increase in power and allow		
C) To enable the engi	ine RPM to be increased with an acco	ompanying increase in propeller RPM.		
21.	A01P	AMP		
Which condition would	d be the least likely to be caused by f	ailed or failing engine bearings?		
A) Excessive oil cons	·			
B) High oil temperatur	B) High oil temperatures.			
C) Low oil temperatur	es.			
22.	A04P	AMP		
What is the basic ope constant speed prope		ower output of an engine equipped with a		
A) Reduce the RPM, then the manifold pressure.				
B) Reduce the manifo	old pressure, then retard the throttle to	o obtain the correct RPM.		
C) Reduce the manifold pressure, then the RPM.				

23.	A04P	AMP	
When will small induction system air leaks have the most noticeable effect on engine operation?			
A) At high RP	M.		
B) At maximul	m continuous and takeoff po	ower settings.	
C) At low RPN	Л.		
24.	A04P	AMP	
	ollowing would most likely c em at low RPM operation?	ause a reciprocating engine to backfire through the	
A) Idle mixture	e too rich.		
B) Clogged de	erichment valve.		
C) Lean mixtu	re.		
25.	A04P	AMP	
Which of the f	ollowing conditions would m	nost likely lead to detonation?	
A) Late ignitio	n timing.		
B) Use of fuel	with too high an octane rati	ng.	
C) Use of fuel	with too low an octane ratir	ıg.	
26.	A04P	AMP	
Which of the f starting the er		perations generally requires engine pre oiling prior to	
A) Engine oil a	and filter change.		
B) Engine inst	allation.		
C) Replaceme	ent of oil lines.		
27.	A04P	AMP	
	er vapor (higher relative hunt It in which of the following?	midity) in the incoming air to a reciprocating engine will	
A) Decreased engine power at a constant RPM and manifold pressure.			
B) Increased p	power output due to increas	ed volumetric efficiency.	
C) A leaning e	effect on engines which use	non automatic carburetors.	
28.	A04P	AMP	
An increase in	n manifold pressure with a co	onstant RPM will cause the bearing load in an engine to	
A) decrease.			
B) remain rela	tively constant.		
C) increase.			

29.	A04P	AMP
•	sh pull carburetor heat control linkage ne diverter valve will be contacted	s should normally be adjusted so that
A) before the stop at	the control lever is reached in both H0	OT and COLD positions.
•	the control lever is reached in the HO ed in the COLD position.	T position and after the stop at the
C) after the stop at th	e control lever is reached in both HOT	Γ and COLD positions.
30.	A04P	AMP
One cause of afterfiring	ng in an aircraft engine is	
A) sticking intake valv	es.	
B) an excessively lea	n mixture.	
C) an excessively rich	n mixture.	
31.	A04P	AMP
To what altitude will a	turbo charged engine maintain sea le	evel pressure?
A) Critical altitude.		
B) Service ceiling.		
C) Pressure altitude.		
32.	A02P	AMP
Master rod bearings a	are generally what type?	
A) Plain.		
B) Roller.		
C) Ball.		
33.	A02P	AMP
Grinding the valves of	f a reciprocating engine to a feather e	dge is likely to result in
A) normal operation a	and long life.	
B) excessive valve cle	earance.	
C) preignition and but	rned valves.	
34.	A02P	AMP
The primary concern	in establishing the firing order for an o	opposed engine is to
A) provide for balance	e and eliminate vibration to the greate	st extent possible.
B) keep power impuls greatest mechanical e	ses on adjacent cylinders as far apart efficiency.	as possible in order to obtain the

	ower impulses on adjacer hanical efficiency.	t cylinders as close as possible in order to obtain the
35.	A02P	AMP
The actual po	ower delivered to the prope	eller of an aircraft engine is called
A) friction hor	sepower.	
B) brake hors	epower.	
C) indicated h	norsepower.	
36.	A02P	AMP
Cam ground	pistons are installed in sor	ne aircraft engines to
A) provide a b	petter fit at operating temp	eratures.
B) act as a co	empensating feature so that	at a compensated magneto is not required.
C) equalize th	ne wear on all pistons.	
37.	A02P	AMP
If the hot clea operation of the		alves when the engine is cold, what will occur during
A) The valves	s will open early and close	early.
B) The valves	s will open late and close of	early.
C) The valves	s will open early and close	late.
38.	A02P	AMP
Full floating pA) the piston.	•	allow motion between the pin and
B) both the pi	ston and the large end of	the connecting rod.
C) both the pi	ston and the small end of	the connecting rod.
39.	A02P	AMP
When is the fo	uel/air mixture ignited in a	conventional reciprocating engine?
A) When the I	piston has reached top de	ad center of the intake stroke.
B) Shortly bef	fore the piston reaches the	e top of the compression stroke.
C) When the	piston reaches top dead c	enter on the compression stroke.
40.	A02P	AMP
On which par occur?	t of the cylinder walls of a	normally operating engine will the greatest amount of wear
A) Near the c	enter of the cylinder where	e piston velocity is greatest.
B) Near the to	op of the cylinder.	

C) Wear is normally of	evenly distributed.	
41.	A02P	AMP
Some cylinder barrel	s are hardened by	
A) nitriding.	·	
B) shot peening.		
C) tempering.		
42.	A02P	AMP
If an engine with a st cylinder will be	roke of 6 inches is operated at 2,000	RPM, the piston movement within the
A) at maximum veloc	ity around TDC.	
B) constant during th	e entire 360° of crankshaft travel.	
C) at maximum veloc	city 90° after TDC.	
43.	A02P	AMP
Some aircraft engine order to	manufacturers equip their product wi	th choked or taper-ground cylinders in
A) provide a straight	cylinder bore at operating temperature	es.
B) flex the rings sligh grooves.	tly during operation and reduce the po	ossibility of the rings sticking in the
C) increase the comp	pression pressure for starting purpose	es.
44.	A02P	AMP
=		engine, and the intake valve opens at 15° on does the intake valve open? (Consider
A) 707°.		
B) 373°.		
C) 347°.		
45.	A02P	AMP
mineral spirits solven	disassembled parts of an engine are trather than water mixed degreasers	, ,
•	s are much more effective.	
	easer residues may cause engine oil of easers cause corrosion.	contamination in the overhauled engine.
46.	A02P	AMP

•	of the cylinder, the volume equals 10 c	•
A) 1:7.		
B) 7:10.		
C) 7:1.		
47.	A02P	AMP
•	rings may be used in chrome plated c	
•	ings may be used in plain steel cylinde	ers.
Regarding the abov		
A) only No. 1 is true		
B) neither No. 1 nor		
C) both No. 1 and N	lo. 2 are true.	
48.	A02P	AMP
	s the ratio between the	
•	he compression stroke and on the inta	ake stroke.
, ·	nber pressure on the combustion strok	
•	with piston at bottom dead center and	
o, o,aoo.a		
49.	B02P	AMP
How does a dual ax	cial flow compressor improve the efficient	ency of a turbojet engine?
A) More turbine who	eels can be used.	
B) Higher compress	sion ratios can be obtained.	
C) The velocity of the	ne air entering the combustion chambe	er is increased.
50.	B02P	AMP
The diffuser section	of a jet engine is located between	
A) the burner sectio	n and the turbine section.	
B) station No. 7 and	station No. 8.	
C) the compressor s	section and the burner section.	
51.	B02P	AMP
Where do stress rup	oture cracks usually appear on turbine	blades?
A) Across the blade	root, parallel to the fir tree.	
B) Along the leading	g edge, parallel to the edge.	
C) Across the leading	ng or trailing edge at a right angle to th	ne edge length

52.	B02P	AMP
An advantage	e of the axial flow compress	sor is its
_	g power requirements.	
B) low weight		
C) high peak		
o) mgm peak	omolorioy.	
53.	B02P	AMP
What is one p	ourpose of the stator blades	s in the compressor section of a turbine engine?
A) Stabilize th	ne pressure of the airflow.	
B) Control the	e direction of the airflow.	
C) Increase the	ne velocity of the airflow.	
54.	B02P	AMP
What is the p	rimary factor which controls	s the pressure ratio of an axial flow compressor?
•	stages in compressor.	р р
•	or inlet pressure.	
•	or inlet temperature.	
C) Compress	or mor tomporataro.	
55.	B02P	AMP
The stator val	nes in an axial flow compre	essor
A) convert ve	locity energy into pressure	energy.
B) convert pre	essure energy into velocity	energy.
C) direct air ir	nto the first stage rotor vand	es at the proper angle.
56.	B02P	AMP
What is the p	roper starting sequence for	a turbojet engine?
A) Ignition, sta		
B) Starter, ign		
C) Starter, fue		
57	B02P	AMD
57.	_	AMP
-	of supersonic air as it now	s through a divergent nozzle
A) increases.		
B) decreases		washa
C) is inversely	proportional to the tempe	rature.
58.	B02P	AMP

What is used in turbin operation?	ne engines to aid in stabilization of cor	npressor airflow during low thrust engine
A) Stator vanes and r	otor vanes.	
•	es and/or compressor bleed valves.	
C) Pressurization and	•	
•	•	
59.	B02P	AMP
The air passing throu	gh the combustion chamber of a turbi	ne engine is
A) used to support co	mbustion and to cool the engine.	
B) entirely combined	with fuel and burned.	
C) speeded up and he	eated by the action of the turbines.	
60.	B02P	AMP
The purpose of a blee urbine engine is to	ed valve, located in the beginning stag	ges of the compressor, in an aircraft gas
A) vent some of the a	ir overboard to prevent a compressor	stall.
B) Control excessively	y high RPM to prevent a compressor	stall.
C) Vent high ram air բ	pressure overboard to prevent a comp	pressor stall.
61.	B02P	AMP
What is the major fun	ction of the turbine assembly in a turb	ojet engine?
A) Directs the gases i	n the proper direction to the tailpipe.	
B) Supplies the powe	r to turn the compressor.	
C) Increases the temp	perature of the exhaust gases.	
62.	B02P	AMP
What type of turbine b	olade is most commonly used in aircra	aft jet engines?
A) Reaction.		
3) Impulse.		
C) Impulse-reaction.		
63.	B02P	AMP
What is the primary a	dvantage of an axial flow compressor	over a centrifugal compressor?
A) High frontal area.		
B) Less expensive.		
C) Greater pressure r	ratio.	
64.	B02P	AMP

embly?
AMP
AMP
tages for both starting flexibility and
AMP
uring turbine engine operation?
AMP
e engine operation?
AMP
e required
d.
AMP
t .

	rotating blades in a turbine engine co iffuse the air. These stationary blades	mpressor, there is a row of stationary are called
A) buckets.		
B) rotors.		
C) stators.		
71.	B02P	AMP
When aircraft turbine expect?	blades are subjected to excessive he	at stress, what type of failures would you
A) Bending and torsic	on.	
B) Torsion and tensio	n.	
C) Stress rupture.		
72.	B02P	AMP
Using standard atmos	spheric conditions, the standard sea le	evel temperature is
A) 59 °F.		
B) 59 °C.		
C) 29 °C.		
73.	B02P	AMP
The velocity of subso	nic air as it flows through a convergen	t nozzle
A) increases.		
B) decreases.		
C) remains constant.		
74.	B03P	AMP
A) allow the turbine w	to shutdown of a turbine engine is actived to cool before the case contracts in the fuel control and/or fuel lines. the engine bearings.	•
75.	B03P	AMP
At what stage in a tur A) Compressor inlet. B) Turbine outlet. C) Compressor outlet	bine engine are gas pressures the gre	eatest?
76.	B03P	AMP

Hot spots in the	ne combustion section of a tu	rbojet engine are possible indicators of
A) faulty ignite	er plugs.	
B) dirty comp	ressor blades.	
C) malfunction	ning fuel nozzles.	
77.	В03Р	AMP
In what section	n of a turbojet engine is the j	et nozzle located?
A) Combustio	n.	
B) Turbine.		
C) Exhaust.		
78.	В03Р	AMP
Newton's Firs	t Law of Motion, generally ter	med the Law of Inertia, states:
A) To every a	ction there is an equal and o	oposite reaction.
B) Force is pr	oportional to the product of n	nass and acceleration.
C) Every body outside force.	· · · ·	or of motion in a straight line, unless acted upon by some
79.	B03P	AMP
If the RPM of be changed b		nains constant, the angle of attack of the rotor blades can
A) changing the	he velocity of the airflow.	
B) changing the	he compressor diameter.	
C) increasing	the pressure ratio.	
80.	В03Р	AMP
-	n the air being introduced int ch of the following?	o the compressor of a turbine engine will form a coating
A) Turbine bla	ades.	
B) Casings.		
C) Inlet guide	vanes.	
81.	B03P	AMP
The Brayton of	cycle is known as the constar	nt
A) pressure c	ycle.	
B) temperatur	e cycle.	
C) mass cycle	9.	
82.	B03P	AMP

The exhaust sec	tion of a turbine engine	is designed to
A) impart a high	exit velocity to the exha	ust gases.
B) increase temp	erature, therefore incre	asing velocity.
C) decrease tem	perature, therefore dec	reasing pressure.
83.	B03P	AMP
	ible cause when a turbi re is high?	ne engine indicates no change in power setting parameters,
B) Engine main b		
,	ge and/or loss of turbin	e efficiency.
84.	B01P	AMP
Who establishes used in general a	•	rating time between overhauls (TBO) of a turbine engine
A) The engine m		
B) The operator (C) The FAA.	(utilizing manufacturer o	data and trend analysis) working in conjunction with the FAA.
85.	B01P	AMP
		engine rotating airfoils does not require special equipment. engine rotating airfoils is commonly recommended by the
Regarding the at	oove statements,	
A) only No. 1 is t	rue.	
B) only No. 2 is t		
C) neither No. 1	nor No. 2 is true.	
86.	B01P	AMP
A) Labyrinth and, B) Teflon and sy	or carbon rubbing.	engines are usually what type(s)?
87.	B01P	AMP
A turbine engine A) double entry c		tains vanes on both sides of the impeller is a

C) single entry	axial flow compressor.	
88.	B01P	AMP
A) perform a fu	Ill power engine run to che the fuel nozzles.	unit has been replaced on an aircraft gas turbine engine? eck fuel flow
89.	B01P	AMP
A) collect the eB) swirl and co	exhaust gases and act as llect the exhaust gases in	
90.	B01P	AMP
When starting	a turbine engine, a hung s	start is indicated if the engine
A) exhaust gas	temperature exceeds sp	ecified limits.
B) fails to reac	h idle RPM.	
C) RPM excee	ds specified operating spe	eed.
91.	B01P	AMP
The blending of	of blades and vanes in a to	urbine engine
A) is usually ac	ccomplished only at engin	e overhaul.
B) should be p points.	erformed parallel to the le	ength of the blade using smooth contours to minimize stress
C) may someti	mes be accomplished wit	h the engine installed, ordinarily using power tools.
92.	B01P	AMP
• .	•	onents exposed to high temperatures may only be marked anufacturer. These materials generally include
1. layout dye.		
2. commercial	felt tip marker.	
3. wax or great	se pencil.	
4. chalk.		
5. graphite lead	d pencil.	
A) 1, 2, and 4.		
B) 1, 3, and 4.		
C) 2, 4, and 5.		

93.	B01P	AMP
The turbine sec	tion of a jet engine	
A) increases ai	r velocity to generate thrust forces	
B) utilizes heat	energy to expand and accelerate	the incoming gas flow.
C) drives the co	ompressor section.	
94.	B01P	AMP
What is the pro	file of a turbine engine compresso	r blade?
A) The leading	edge of the blade.	
B) A cutout that	reduces blade tip thickness.	
C) The curvatu	re of the blade root.	
95.	B01P	AMP
The abbreviation	on P _{t7} used in turbine engine termi	nology means
A) the total inle	t pressure.	
B) pressure and	d temperature at station No. 7.	
C) the total pres	ssure at station No. 7.	
96.	B01P	AMP
Which stateme	nt is true regarding jet engines?	
•	•	apidly with small increases in RPM.
B) At the higher	r engine speeds, thrust increases	rapidly with small increases in RPM.
C) The thrust d	elivered per pound of air consume	d is less at high altitude than at low altitude.
97.	B01P	AMP
Turbine nozzle turbine engine		am side of each turbine wheel, are used in the gas
A) decrease the	e velocity of the heated gases flow	ing past this point.
B) direct the flo	w of gases parallel to the vertical	ine of the turbine blades.
C) increase the	velocity of the heated gases flow	ng past this point.
98.	B01P	AMP
An exhaust cor the exhaust due	-	engine will cause the pressure in the first part of
A) increase and	the velocity to decrease.	
B) increase and	d the velocity to increase.	
C) decrease an	d the velocity to increase.	

99.	B01P	AMP
One function of t	he nozzle diaphragm in a tur	bine engine is to
A) decrease the	velocity of exhaust gases.	
B) center the fue	el spray in the combustion cha	amber.
C) direct the flow	of gases to strike the turbine	e blades at the desired angle.
100.	C01P	AMP
Serviceability vanes.	limits for turbine blades are	much more stringent than are those for turbine nozzle
(2) A limited num blade.	nber of small nicks and dents	can usually be permitted in any area of a turbine
Regarding the al	bove statements,	
A) both No. 1 an	d No. 2 are true.	
B) neither No. 1	nor No. 2 is true.	
C) only No. 1 is t	true.	
101.	C01P	AMP
What section in t	the instructions for continued	airworthiness is FAA approved?
A) Engine mainte	enance manual or section.	
B) Engine overh	aul manual or section.	
C) Airworthiness	limitations section.	
102.	C01P	AMP
What publication	contains the mandatory repl	acement time for parts of a turbine engine?
	facturer's service instructions	
B) Engine Manut	facturer's maintenance manu	al.
C) Airworthiness	directive issued by the engir	ne manufacturer.
103.	C01P	AMP
		ollowing a major repair of an aircraft engine?
		d a list of discrepancies for the FAA.
,	engine maintenance record a	·
C) Entry in logbo	•	
104.	C01P	AMP
		ype certificates for small airplanes with nine or less
		obatic categories may be found in the
A) Supplemental	Type Certificate.	

,	Regulations, Part 23. Regulations, Part 21.	
o) i caciai Aviation	rtogulations, i art 21.	
105.	C01P	AMP
Which of the following	ng can inspect and approve	e an engine major repair for return to service?
,	hanic with airframe and pov	
,	hanic with a powerplant rati	•
C) Certificated meci	hanic with inspection autho	rization.
106.	C01P	AMP
	iting conditions of a turbine ure. sperature.	echanical integrity of the turbines, as well as to engine?
107.	C01P	AMP
•	nt figure 1.) Determine whic I No. 5863-40 with 283 hou	h portion of the AD is applicable for Model O-690 rs` time in service.
108.	C01P	AMP
Straightening nitride		
A) recommended.		
B) not recommende	ed.	
C) approved by the	manufacturer.	
109.	C01P	AMP
You are performing indicate?	a 100-hour inspection on a	in R985-22 aircraft engine. What does the '985'
A) The total piston o	displacement of the engine.	
B) The pistons will p	oump a maximum of 985 cu	bic inches of air per crankshaft revolution.
C) The total piston of	displacement of one cylinde	r.
110.	C01P	AMP
	ng contains a table that list rtificate Data Sheets.	s the engines to which a given propeller is adaptable?

B) Propeller Type Ce	rtificate Data Sheets.	
C) Engine Type Certi	ficate Data Sheets.	
111.	C01P	AMP
Which of the following	g contains a minimum	checklist for 100-hour inspections of engines?
A) 14 CFR Part 33 Ap	ppendix A.	
B) 14 CFR Part 43 Ap	ppendix D.	
C) Engine Specification	ons or Type Certificat	e Data Sheets.
112.	C01P	AMP
		lity statement which applies to an IVO 355 engine, serial and 300 hours since rebuilding.
A) Applies to all IVO and time.	355 engines, serial nu	mbers T8000 through T8300, having less than 2,400
B) Applies to all IVO : total time.	355 engines, serial nu	mbers T8000 through T8900 with 2,400 hours or more
C) Applies to all I.O. aoverhaul.	and TV10-355 engine	s, all serial numbers regardless of total time or since
113.	C01P	AMP
damaged in a ground	l accident, and this mo	peller Model No. 2A34C50/90A. The propeller is severely del propeller is not available for replacement. Which of roved alternate replacement?
A) Summary of Suppl	lemental Type Certific	ates.
B) Aircraft Specification	ons/Type Certificate D	ata Sheets.
C) Aircraft Engine and	d Propeller Specificati	ons/ Type Certificate Data Sheets.
114.	H02P	AMP
indication on a turbos	gulated power change super-charged aircraft	s that result in continual drift of manifold pressure engine is known as
A) Overshoot.		
B) Waste gate fluctua	ation.	
C) Bootstrapping.		
115.	H02P	AMP
A Bourdon tube instru	ument may be used to	indicate
1. pressure.		
2. temperature.		
3. position.		

4. quantity.		
A) 1 and 2.		
B) 1 and 3.		
C) 2 and 4.		
116.	H02P	AMP
Which of the following instr	rument discrepancies require replac	ement of the instrument?
1. Red line missing from gla	ass.	
2. Glass cracked.		
Case paint chipped.		
4. Will not zero out.		
5. Pointer loose on shaft.		
6. Mounting screw loose.		
7. Leaking at line B nut.		
8. Fogged.		
A) 2, 3, 7, 8.		
B) 2, 4, 5, 8.		
C) 1, 2, 4, 7.		
117.	H02P	AMP
Instruments that measure r what type?	elatively high fluid pressures, such	as oil pressure gauges, are usually
A) Vane with calibrated spr	ing.	
B) Bourdon tube.		
C) Diaphragm or bellows.		
118.	H02P	AMP
In what units are turbine en	ngine tachometers calibrated?	
A) Percent of engine RPM.		
B) Actual engine RPM.		
C) Percent of engine press	ure ratio.	
119.	H02P	AMP
What would be the possible fuel flow, and low RPM at a		high exhaust gas temperature, high
A) Fuel control out of adjus	tment.	
B) Loose or corroded therm	nocouple probes for the EGT indica	tor.
C) Turbine damage or loss	of turbine efficiency.	

120.	H02P	AMP
In regard to using	a turbine engine oil analys	is program, which of the following is NOT true?
A) Generally, an a	accurate trend forecast may	be made after an engine's first oil sample analysis.
B) It is best to sta	rt an oil analysis program o	n an engine when it is new.
C) A successful on normal trends car		pe run over an engine's total operating life so that
121.	H02P	AMP
A red triangle, do	t, or diamond mark on an er	ngine instrument face or glass indicates
A) the maximum	operating limit for all normal	operations.
B) the maximum	limit for high transients such	as starting.
C) a restricted op	erating range.	
122.	H02P	AMP
The EGT gauge ι order to	used with reciprocating engi	nes is primarily used to furnish temperature readings in
A) obtain the best	t mixture setting for fuel effic	ciency.
B) obtain the best	t mixture setting for engine	cooling.
C) prevent engine	e overtemperature.	
123.	H02P	AMP
On an aircraft turb result in	bine engine, operating at a	constant power, the application of engine anti-icing will
A) noticeable shif	t in EPR.	
B) a false EPR re	ading.	
C) an increase in	EPR.	
124.	H02P	AMP
Which of the follo	wing is a primary engine ins	strument?
A) Tachometer.		
B) Fuel flowmeter	r.	
C) Airspeed indic	ator.	
125.	H02P	AMP
Which statement system?	is true regarding a thermoo	ouple type cylinder head temperature measuring
A) The resistance	e required for cylinder head	temperature indicators is measured in farads.

,	nds of the thermocouple.	ir is determined by the temperature difference
C) When the maste	er switch is turned on, a ther	mocouple indicator will move off scale to the low side
126.	H02P	AMP
Basically, the indic	ator of a tachometer system	is responsive to change in
A) current flow.		
B) frequency.		
C) voltage.		
127.	H02P	AMP
Which of the follow	ring types of electric motors a	are commonly used in electric tachometers?
A) Direct current, s	series wound motors.	
B) Synchronous m	otors.	
C) Direct current, s	shunt-wound motors.	
128.	H02P	AMP
A manifold pressur	e gauge is designed to	
A) maintain consta	nt pressure in the intake ma	nifold.
B) indicate differen	itial pressure between the int	ake manifold and atmospheric pressure.
C) indicate absolut	e pressure in the intake mar	ifold.
129.	H02P	AMP
•	trument range markings show ble for a limited time, or unau	w whether the current state of powerplant operation athorized.
•		pased on installed engine operating limits which may ose limits shown on the engine Type Certificate
Regarding the abo	ve statements,	
A) both No. 1 and I	No. 2 are true.	
B) neither No. 1 no	or No. 2 is true.	
C) only No. 1 is tru	e.	
130.	H02P	AMP
` ' • .	e ratio (EPR) is a ratio of the cates the thrust produced.	exhaust gas pressure to the engine inlet air
` ' • •	e ratio (EPR) is a ratio of the cates volumetric efficiency.	exhaust gas pressure to the engine inlet air

Regarding the above statements,

A) only No. 1 is tr	ue.	
B) only No. 2 is tr	ue.	
C) both No. 1 and	l No. 2 are true.	
131.	H02P	AMP
Thermocouple lea	ads	
A) may be installed	ed with either lead to either p	ost of the indicator.
B) are designed for	or a specific installation and	may not be altered.
C) may be repaire	ed using solderless connecto	rs.
132.	H02P	AMP
temperature gaug	ge pointer indicate?	crossed at installation, what would the cylinder
,	rature for prevailing conditior	
B) Moves off scal	e on the zero side of the met	er.
C) Moves off scal	e on the high side of the met	er.
133.	H02P	AMP
significant damag	e), it means that the engine	
(2) Some turbine EPR as the prima	-	s the primary indicator of thrust produced, others use
Regarding the ab	ove statements,	
A) only No. 1 is tr	ue.	
B) only No. 2 is tr	ue.	
C) both No. 1 and	l No. 2 are true.	
134.	H01P	AMP
flow transmitters i	s a measure of	driven impeller and turbine, and motorless type fuel
A) fuel mass flow		
B) fuel volume flo		
C) engine burner	pressure drop.	
135.	H01P	AMP
The fuel flow indic measure of	cation system used with man	y fuel-injected opposed engine airplanes utilizes a
A) fuel flow volum	ne.	
B) fuel pressure.		

C) fuel flow mass	3.	
136.	H01P	AMP
Motor driven imp	eller and turbine fuel flow tran	smitters are designed to transmit data
A) using aircraft e	electrical system power.	
B) mechanically.		
C) by fuel pressu	re.	
137.	H01P	AMP
opposed reciprod	cating engines measures the f	injection system installed on an aircraft horizontally uel pressure drop across the
A) manifold valve).	
B) fuel nozzles.		
C) metering valve) .	
138.	I01P	AMP
	e line exit ports are covered w	wo small lines running from the system and exiting ith a blowout type indicator disc. Which of the
A) When the red normally discharg	_	dicates the fire extinguishing system has been
B) When the yellonormally discharge	_	indicates the fire extinguishing system has been
C) When the gree thermal discharge	<u> </u>	indicates the fire extinguishing system has had a
139.	I01P	AMP
Which of the follotemperature?	owing fire detection systems n	neasures temperature rise compared to a reference
A) Thermocouple).	
B) Thermal switch	h.	
C) Lindberg conti	inuous element.	
140.	I01P	AMP
How are most air	craft turbine engine fire exting	juishing systems activated?
A) Electrically dis	charged cartridges.	
B) Manual remote	e control valve.	
C) Pushrod asse	mbly.	

141.	I01P	AMP
Why does one type of Fenw two separate circuits?	ral fire detection system use spo	ot detectors wired in parallel between
A) To provide an installation secondary, or back-up system	-	ystems: a primary system and a
B) So that a double fault ma	y exist in the system without so	unding a false alarm.
C) So that a single fault may	exist in the system without sou	unding a false alarm
142.	I01P	AMP
A fuel or oil fire is defined as	s a	
A) class B fire.		
B) class A fire.		
C) class C fire.		
143.	I01P	AMP
The explosive cartridge in the	ne discharge valve of a fire extin	nguisher container is
A) a life dated unit.		
B) not a life dated unit.		
C) mechanically fired.		
144.	I01P	AMP
_	juishing agent for a carburetor of	or intake fire is
A) carbon dioxide.		
B) dry chemical.C) methyl bromide.		
C) metryr bronnide.		
145.	I01P	AMP
(Refer to Powerplant figure temperature is 75 °F.	Determine the fire extinguish	ner container pressure limits when the
A) 326 minimum and 415 m	aximum.	
B) 330 minimum and 419 m	aximum.	
C) 338 minimum and 424 m	aximum.	
146.	I01P	AMP
The fire detection system th beads in a tube is the	at uses a single wire surrounde	d by a continuous string of ceramic
A) Fenwal system.		
B) Kidde system.		
C) thermocouple system.		

147.	I01P	AMP
A continuous loo	p fire detector is what type of c	letector?
A) Spot detector.		
B) Overheat dete	ector.	
C) Rate of tempe	erature rise detector.	
148.	I01P	AMP
	nguished, or overheat conditio ctor, the detection system	n removed in aircraft equipped with a Systron-
A) must be manu	ıally reset.	
B) automatically	resets.	
C) sensing comp	onent must be replaced.	
149.	I01P	AMP
What is the funct	ion of a fire detection system?	
A) To discharge	the powerplant fire extinguishir	ng system at the origin of the fire.
B) To activate a	warning device in the event of	a powerplant fire.
C) To identify the	e location of a powerplant fire.	
150.	I01P	AMP
What retains the container?	nitrogen charge and fire exting	guishing agent in a high rate of discharge (HRD)
A) Breakable dis	k and fusible disk.	
B) Pressure swite	ch and check tee valve.	
C) Pressure gau	ge and cartridge.	
151.	I01P	AMP
	or down) of an illuminated fire help accomplishes what events?	nandle in a typical large jet aircraft fire protection
A) Closes all fire	wall shutoff valves, disconnect	s the generator, and discharges a fire bottle.
B) Closes fuel shextinguishing sys	_	disconnects the generator field, and arms the fire
•	nutoff, closes hydraulic shutoff, and arms the fire-extinguishing	closes the oxygen shutoff, disconnects the system.
152.	I01P	AMP
	owing fire detectors are commo	only used in the power section of an engine nacelle?

B) Smoke detector	rs.	
C) Rate of temperation	ature rise detectors.	
153.	J02P	AMP
When selecting an	n electrical switch for install	ation in an aircraft circuit utilizing a direct current motor,
	ed for dc should be chose	_
B) a derating factor	or should be applied.	
C) only switches w	vith screw type terminal co	nnections should be used.
154.	J02P	AMP
engine fails to rota A) power lever swi	ite, one of the probable car itch is defective.	
,	plenoid contacts are defect	ive.
C) starter solenoid	I is defective.	
155.	J02P	AMP
The maximum allo	wable voltage drop between	en the generator and the bus bar is
A) 1 percent of the	e regulated voltage.	
B) 2 percent of the	e regulated voltage.	
C) less than the vo	oltage drop permitted betw	een the battery and the bus bar.
156.	J02P	AMP
Bonding jumpers s	should be designed and ins	stalled in such a manner that they
A) are not subjecte	ed to flexing by relative mo	tion of airframe or engine components.
B) provide a low e	lectrical resistance in the g	round circuit.
C) prevent buildup atmosphere.	of a static electrical charg	e between the airframe and the surrounding
157.	J02P	AMP
•	,	rstem, what is the maximum continuous current that 25 feet long, routed in free air?
A) 20 amperes.		
B) 35 amperes.		
C) 28 amperes.		
158.	J02P	AMP
When installing ele	ectrical wiring parallel to a	fuel line, the wiring should be
A) in metal condui	t.	

B) in a non-conductive fiC) above the fuel line.	ire-resistant sleeve.	
159.	J02P	AMP
What type of lubricant mA) Silicone grease.B) Soapstone talc.C) Rubber lubricant.	nay be used to aid in p	oulling electrical wires or cables through conduits?
160.	J02P	AMP
Which Federal Aviation breakers? A) 14 CFR Part 21. B) 14 CFR Part 23. C) 14 CFR Part 91.	Regulation requireme	ent prevents the use of automatic reset circuit
161.	J02P	AMP
yrnat is the maximum not grounded to a flat surfact A) Two. B) Three. C) Four.		per wires that may be attached to one terminal
162.	J02P	AMP
Which of the following is A) Speed of the armatur B) Number of windings is C) The strength of the field	re. in the armature.	ator to control its voltage output?
163.	J02P	AMP
(1) Electrical circuit prote without overheating the		ed based on the amount of current that can be carried
(2) A 'trip free' circuit bre excessive current is flow		sible to manually hold the circuit closed when
Regarding the above sta	atements,	
A) only No. 1 is true.		
B) only No. 2 is true.		
C) both No. 1 and No. 2	are true.	

164.	J02P	AMP
What is the smallest A) No. 6. B) No. 8. C) No. 10.	terminal stud allowed	for aircraft electrical power systems?
165. A term commonly us terminal strip is A) strapping. B) stepping. C) stacking.	J02P sed when two or more o	AMP electrical terminals are installed on a single lug of a
A) Continually, as lo B) Continually, as lo	•	
167. Aircraft electrical wir A) Military Specifica B) American Wire G C) Technical Standa	auge system.	AMP cording to the
168. Aircraft copper elect A) improve conducti B) add strength. C) prevent oxidizatio	vity.	AMP n tin, silver, or nickel in order to
169. As a general rule, st A) one half their orig B) one-third their ori C) two-thirds their or	inal length. ginal length.	AMP ced when they are approximately
170.	J01P	AMP

Alternators (ac gene regulate the alternate	,	constant-speed drive (CSD) mechanism are used to	,
A) voltage output.			
B) amperage output.			
C) hertz output.			
171.	J01P	AMP	
A high surge of curremotor increases,	ent is required when a dc el	ectric motor is first started. As the speed of the	
A) the counter emf d	lecreases proportionally.		
B) the applied emf in	ncreases proportionally.		
C) the counter emf b the armature.	ouilds up and opposes the a	pplied emf, thus reducing the current flow through	
172.	J01P	AMP	
If a generator is malf A) rheostat.	functioning, its voltage can	be reduced to residual by actuating the	
B) generator master	switch.		
C) master solenoid.			
173.	J01P	AMP	
	often classified according to agine starter motors are ger	the method of connecting the field coils and erally of which type?	
C) Shunt (parallel).			
174.	J01P	AMP	
The generating syste	em of an aircraft charges th	e battery by using	
A) constant current a	and varying voltage.		
B) constant voltage a	and varying current.		
C) constant voltage	and constant current.		
175.	J01P	AMP	
What is the frequenc	cy of most aircraft alternatin	g current?	
A) 115 Hertz.			
B) 60 Hertz.			
C) 400 Hertz.			

176.	J01P	AMP	
What are two types of ac mo	otors that are used to produce a re	elatively high torque?	
A) Shaded pole and shunt field.			
B) Shunt field and single pha	ase.		
C) Three phase induction ar	nd capacitor start.		
	–		
177.	J01P	AMP	
Generator voltage will not but plate. These are most likely		nd solder is found on the brush cover	
A) an open armature.			
B) excessive brush arcing.			
C) armature shaft bearings of	overheating.		
178.	J01P	AMP	
Why is it unnecessary to flas	sh the field of the exciter on a brus	shless alternator?	
A) The exciter is constantly	charged by battery voltage.		
B) Brushless alternators do	not have exciters.		
C) Permanent magnets are	installed in the main field poles.		
470	10.4 D	4140	
179.	J01P	AMP	
•		ted on gas turbine engines is by a	
A) drop in compressor disch	•		
B) sensing switch located in	the tallpipe.		
C) drop in fuel flow.			
180.	J01P	AMP	
If the points in a vibrator type operating, what will be the p		sed position while the generator is	
A) Generator output voltage	will decrease.		
B) Generator output voltage	will not be affected.		
C) Generator output voltage	will increase.		
181.	J01P	AMP	
What is a basic advantage of	of using ac for electrical power for	a large aircraft?	
_	gher voltage than dc systems and	d therefore use less current and can	
-	wer voltage than dc systems and	therefore use less current and can	

use smaller and ligh	0	in ac systems and therefore use more current and can
182.	J01P	AMP
•	-	of electricity, when a properly functioning dc alternator a aircraft's battery, the direction of current flow through
A) is into the negative	e terminal and out the p	ositive terminal.
B) is into the positive	e terminal and out the ne	gative terminal.
C) cycles back and to speed of the alternation		cycles per second being controlled by the rotational
183.	J01P	AMP
A) So that the voltage B) To eliminate unco	ge output of the generato ontrolled surges of curre	ol the speed of some aircraft engine driven generators? or will remain within limits. Int to the electrical system. International constant.
184.	J01P	AMP
The stationary field s A) by the reverse-cu B) because of gener C) according to the I	rator speed.	nt generator is varied
185.	J01P	AMP
	to convert alternating contraction and convert alternating contractions and convert alternating contractions.	urrent, which has been induced into the loops of the current?
186.	K01P	AMP
What will be the restrection of the commended by the A) The oil pressure vB) The oil temperature.	-	e in extremely high temperatures using a lubricant ch lower temperature? al. be higher than normal.
187.	K01P	AMP

Upon what quality or cha	aracteristic of a lu	bricating oil is its viscosity index based?
A) Its resistance to flow the same temperature.	at a standard ten	perature as compared to high grade paraffin base oil at
B) Its rate of change in v	viscosity with tem	perature change.
C) Its rate of flow throug	h an orifice at a s	tandard temperature.
188.	K01P	AMP
The oil used in reciproca	ating engines has	a relatively high viscosity due to
A) the reduced ability of pressure).	thin oils to maint	ain adequate film strength at altitude (reduced atmospheric
B) the relatively high rota	ational speeds.	
C) large clearances and	high operating to	mperatures.
189.	K01P	AMP
In addition to lubricating functions?	(reducing friction	between moving parts), engine oil performs what
1. Cools.		
2. Seals.		
3. Cleans.		
4. Prevents corrosion.		
5. Cushions impact (sho	ock) loads.	
A) 1, 2, 3, 4.		
B) 1, 2, 3, 4, 5.		
C) 1, 3, 4.		
190.	K01P	AMP
Which of the following fa	actors helps deter	mine the proper grade of oil to use in a particular engine?
A) Adequate lubrication	in various attitude	es of flight.
B) Positive introduction	of oil to the bearin	ngs.
C) Operating speeds of	bearings.	
191.	K01P	AMP
	d high rubbing ve	elocities, such as occur with spur type gears, require the
A) an EP lubricant.		
B) straight mineral oil.		
C) metallic ash deterger	nt oil.	
192.	K01P	AMP

vvnat type of oli de	o most engine manufacturers	recommend for new reciprocating engine break in?
A) Ashless disper	sant oil.	
B) Straight minera	al oil.	
C) Semi synthetic	oil.	
193.	K03P	AMP
How are the teeth	of the gears in the accessor	y section of an engine normally lubricated?
A) By splashed or	sprayed oil.	
B) By submerging	the load bearing portions in	oil.
C) By surrounding be maintained.	g the load bearing portions wi	th baffles or housings within which oil pressure can
194.	K03P	AMP
If the oil in the oil the cooler?	cooler core and annular jack	et becomes congealed, what unit prevents damage to
A) Oil pressure re	lief valve.	
B) Airflow control	valve.	
C) Surge protection	on valve.	
195.	K03P	AMP
What will result if	an oil filter becomes complet	ely blocked?
A) Oil will flow at a	a reduced rate through the sy	rstem.
B) Oil flow to the	engine will stop.	
C) Oil will flow at t	the normal rate through the s	ystem.
196.	K03P	AMP
The vent line conr permits	necting the oil supply tank an	d the engine in some dry sump engine installations
A) pressurization	of the oil supply to prevent ca	avitation of the oil supply pump.
B) oil vapors from	the engine to be condensed	and drained into the oil supply tank.
C) the oil tank to b	pe vented through the norma	engine vent.
197.	K03P	AMP
		the clearances between the moving parts of an
engine increase th	nrough normal wear, the supp	oly pump output
A) increases as th	ne resistance offered to the flo	ow of oil increases.
B) remains relativerelief valve.	ely constant (at a given RPM) with less oil being returned to the pump inlet by the

C) remains relative the relief valve.	vely constant (at a given RPM)	with more oil being returned to the pump inlet by
198.	K03P	AMP
An engine lubrica	tion system pressure relief val	ve is usually located between the
A) oil cooler and t	the scavenger pump.	
B) scavenger pun	np and the external oil system	
C) pump and the	internal oil system.	
199.	K03P	AMP
A drop in oil press	sure may be caused by	
A) the temperatur	re regulator sticking open.	
B) the bypass val	ve sticking open.	
C) foreign materia	al under the relief valve.	
200.	K03P	AMP
Which type valve running?	prevents oil from entering the	main accessory case when the engine is not
A) Bypass.		
B) Relief.		
C) Check.		
201.	K03P	AMP
As a general rule, magnetic chip de		y particles or gray metallic paste on a turbine engine
A) is considered t	to be the result of normal wear	•
B) indicates an im	nminent component failure.	
C) indicates acce	lerated generalized wear.	
202.	K03P	AMP
The purpose of a	dwell chamber in a turbine en	gine oil tank is to provide
A) a collection po	int for sediments.	
B) for a pressuriz	ed oil supply to the oil pump ir	ılet.
C) separation of e	entrained air from scavenged o	pil.
203.	K03P	AMP
The purpose of a	relief valve installed in the tan	k venting system of a turbine engine oil tank is to
A) prevent oil pun	np cavitation by maintaining a	constant pressure on the oil pump inlet.

B) maintain internal ta of change in altitude.	ank air pressure at the an	nbient atmospheric level regardless of altitude or rate
_	internal pressure in the	oil tank after shutdown to prevent oil pump cavitation
204.	K03P	AMP
From the following, id engine.	entify the factor that has	the least effect on the oil consumption of a specific
A) Mechanical efficier	ncy.	
B) Engine RPM.		
C) Lubricant characte	ristics.	
205.	K03P	AMP
How is the oil collecte	ed by the piston oil ring re	eturned to the crankcase?
B) Through holes drill	s cut in the piston wall be led in the piston oil ring g led in the piston pin reces	
206.	K03P	AMP
		ion system thins the oil with
A) kerosene.	and the same of th	
B) alcohol.		
C) gasoline.		
207.	K03P	AMP
Where is the oil tempo	erature bulb located on a	a dry sump reciprocating engine?
A) Oil inlet line.		
B) Oil cooler.		
C) Oil outlet line.		
208.	K03P	AMP
If a full flow oil filter is	used on an aircraft engir	ne, and the filter becomes completely clogged, the
A) oil supply to the en	igine will be blocked.	
	d back to the oil tank hop sage through the engine.	oper where larger sediments and foreign matter will
C) bypass valve will o	pen and the oil pump wil	Il supply unfiltered oil to the engine.
209.	K03P	AMP

	the cylinders of an inverted i reduced or prevented by	in line engine and in the lower cylinders of a radial
A) reversed oil con	·	
,		ating oil to a separate scavenger pump.
C) extended cylinder	•	
210.	K03P	AMP
Why is an aircraft rovent line?	eciprocating engine oil tank o	on a dry sump lubrication system equipped with a
A) To prevent pres	sure buildup in the reciprocat	ing engine crankcase.
B) To eliminate foa	ming in the oil tank.	
C) To prevent pres	sure buildup in the oil tank.	
211.	K03P	AMP
The purpose of the	flow control valve in a recipr	ocating engine oil system is to
A) direct oil through	n or around the oil cooler.	
B) deliver cold oil to	o the hopper tank.	
C) compensate for	volumetric increases due to	foaming of the oil.
212.	K03P	AMP
What determines the stacked disc, edge	•	ch will be excluded or filtered by a cuno type
A) The disc thickne	ess.	
B) The spacer thick	rness.	
C) Both the numbe	r and thickness of the discs in	n the assembly.
213.	K03P	AMP
The pumping capa	city of the scavenger pump ir	a dry sump aircraft engine's lubrication system
A) is greater than th	he capacity of the oil supply p	oump.
B) is less than the	capacity of the oil supply pur	ıp.
C) is usually equal conditions.	to the capacity of the oil supp	oly pump in order to maintain constant oiling
214.	K02P	AMP
Oil picks up the mo	est heat from which of the follows:	owing turbine engine components?
A) Rotor coupling.		
B) Compressor bea	aring.	
C) Turbine bearing		

215.	K02P	AMP
•	pressure within the lubricating of ocating engine)?	il tank from rising above or falling below ambient
A) Oil tank chec	k valve.	
B) Oil pressure	relief valve.	
C) Oil tank vent		
216.	K02P	AMP
What is the purp	oose of the last chance oil filters	?
A) To prevent da	amage to the oil spray nozzle.	
B) To filter the o	il immediately before it enters th	ne main bearings.
C) To assure a	clean supply of oil to the lubrica	tion system.
217.	K02P	AMP
In a reciprocatin	g engine oil system, the temper	ature bulb senses oil temperature
A) at a point after	er the oil has passed through the	e oil cooler.
B) while the oil i	s in the hottest area of the engi	ne.
C) immediately	before the oil enters the oil cool	er.
218.	K02P	AMP
	some oil will flow through the re elief valve is set at a pressure w	lief valve of a gear type engine oil pump. This is hich is
A) lower than th	e pump inlet pressure.	
B) lower than th	e pressure pump capabilities.	
C) higher than p	ressure pump capabilities.	
219.	K02P	AMP
The oil dampen	ed main bearing utilized in some	e turbine engines is used to
A) provide lubrio established.	cation of bearings from the begin	nning of starting rotation until normal oil pressure is
<i>,</i> .	I film between the outer race an e rotor system, and to allow for	d the bearing housing in order to reduce vibration slight misalignment.
C) dampen surg	ges in oil pressure to the bearing	ys.
220.	K02P	AMP
The engine oil to sump reciprocate		located between which of the following on a dry
A) The engine of	il supply pump and the internal	lubrication system.
B) The scaveng	er pump outlet and the oil storage	ge tank.

C) The oil storage tank ar	nd the engine oil supply pump.	
221.	K02P	AMP
Possible failure related fe type magnetic chip detec A) disturbing the magneti B) bridging the gap betwe	tor to indicate their presence by c lines of flux around the detector tien the detector center (positive) electric current that is caused by the p	ectrode and the ground electrode.
222.	K02P	AMP
What is the primary purpond. A) Cool the fuel. B) Cool the oil. C) De aerate the oil.	ose of the oil to fuel heat exchanger	?
223.	K02P	AMP
A) should be limited to the B) has a negligible effect. C) will not occur because 224.	e engine manufacturer's recommen	rings.
A) positive displacement.B) variable displacement.C) constant speed.	·	
225.	L02P	AMP
A) The distributor turns at B) The distributor turns at	etween distributor and crankshaft some half crankshaft speed. I one and one half crankshaft speed. It one half distributor speed.	speed of aircraft reciprocating engines?
226. Which of the following are 1. Magnetic. 2. Primary.	L02P e distinct circuits of a high tension n	AMP nagneto?

3. E gap.4. P lead.5. Secondary.A) 1, 2, 5.B) 1, 3, 4.C) 2, 4, 5.			
227.	L02P		AMP
Which of the following at 1. Gives a more complet 2. Provides a backup ma 3. Increases the output p 4. Permits the use of low 5. Increases the intensity A) 2, 3, 4. B) 2, 3, 5. C) 1, 2, 3.	te and quick combagneto system. bower of the engineer grade fuels.	oustion of the fuel.	oft engines?
228.	L02P		AMP
Ignition check during end A) defective spark plugs B) a defective high tensi C) incorrect ignition timin	on lead.		PM. This is usually caused by
229.	L02P		AMP
Using a cold spark plug A) normal operation. B) a fouled plug. C) detonation.		ssion aircraft engine	
230.	L02P		AMP
magneto. The major por position (fast drop). The A) faulty or fouled spark B) incorrect ignition timin	tion of the RPM lo most likely cause plugs. ng on both magne	oss occurs rapidly aft e is	drop during operation on the right ter switching to the right magneto
C) one or more dead cyl	inders.		

231.	L02P	AMP
Upon inspection of the black soot. This indica		rcraft engine, the plugs were found caked with a heavy
A) worn oil seal rings.		
B) a rich mixture.		
C) a lean mixture.		
232.	L02P	AMP
Which of the following	would be cause for r	ejection of a spark plug?
A) Carbon fouling of th	e electrode and insu	lator.
B) Insulator tip cracked	d.	
C) Lead fouling of the	electrode and insulat	tor.
233.	L02P	AMP
Defective spark plugs	will cause the engine	to run rough at
A) high speeds only.		
B) low speeds only.		
C) all speeds.		
234.	L02P	AMP
The type of ignition sys	stem used on most to	urbine aircraft engines is
A) high resistance.		
B) low tension.		
C) capacitor discharge	·.	
235.	L02P	AMP
Which of the following	statements most acc	curately describes spark plug heat range?
A) The length of the th	readed portion of the	shell usually denotes the spark plug heat range.
,		tor tip is reasonably short to hasten the rate of heat shell to the cylinder head.
		ator tip is reasonably short to hasten the rate of heat shell to the cylinder head.
236.	L02P	AMP
The secondary coil of	a magneto is ground	ed through the
A) ignition switch.		
B) primary coil.		
C) grounded side of th	e breaker points.	

237.	L02P	AMP
When removing a shi	elded spark plug, whic	h of the following is most likely to be damaged?
A) Center electrode.		
B) Shell section.		
C) Core insulator.		
238.	L02P	AMP
	nition system is used a	Imost universally on turbine engines primarily because
A) low amperage. B) long life.		
C) high heat intensity		
239.	L02P	AMP
In a turbine engine do formed?	capacitor discharge i	gnition system, where are the high voltage pulses
A) At the breaker.		
B) At the triggering tra	ansformer.	
C) At the rectifier.		
240.	L02P	AMP
Which of the following	g statements regarding	magneto switch circuits is NOT true?
A) In the BOTH position	on, the right and left m	agneto circuits are grounded.
B) In the OFF position	n, neither the right nor	left magneto circuits are open.
C) In the RIGHT posit	tion, the right magneto	circuit is open and the left magneto circuit is grounded.
241.	L02P	AMP
Hot spark plugs are g	enerally used in aircra	ft powerplants
A) with comparatively	high compression or I	nigh operating temperatures.
B) with comparatively	low operating temperating	atures.
C) which produce high	h power per cubic inch	displacement.
242.	L02P	AMP
The term 'reach,' as a	pplied to spark plug d	esign and/or type, indicates the
A) linear distance fror	n the shell gasket sea	to the end of the threads on the shell skirt.
B) length of center ele	ectrode exposed to the	flame of combustion.
C) length of the shield	ded barrel.	

243.	L02P	AMP	
A spark plug's h	eat range is the result of		
A) the area of th	e plug exposed to the cooling a	airstream.	
3) its ability to tr	ansfer heat from the firing end	of the spark plug to the cylinder head.	
C) the heat inter	nsity of the spark.		
244.	L02P	AMP	
	owing, obtained during magnet he right magneto primary and t	o check at 1,700 RPM, indicates a short (grounde the ignition switch?	d)
A) BOTH-1,700	RPM; R-1,625 RPM; L-1,700 F	PM; OFF-1,625 RPM.	
B) BOTH-1,700	RPM; R-0 RPM; L-1,700 RPM;	OFF-0 RPM.	
C) BOTH-1,700	RPM; R-0 RPM; L-1,675 RPM;	OFF-0 RPM.	
245.	L02P	AMP	
Sharp bends sho	ould be avoided in ignition lead	s primarily because	
A) weak points r	nay develop in the insulation th	rough which high tension current can leak.	
3) ignition lead v	wire conductor material is brittle	and may break.	
C) ignition lead s	shielding effectiveness will be r	educed.	
246.	L02P	AMP	
Spark plugs are	considered worn out when the		
	ive worn away to about one-ha		
3) center electro	ode edges have become round	ed.	
C) electrodes ha	eve worn away to about two-thing	rds of their original dimensions.	
247.	L02P	AMP	
	ed in turbine engines are subject vice life because they	cted to high intensity spark discharges and yet the	y
A) operate at mu	uch lower temperatures.		
3) are not place	d directly into the combustion o	hamber.	
C) do not require	e continuous operation.		
248.	L02P	AMP	
•	removing a turbine engine ign ving a lethal shock, the ignition	iter plug, in order to eliminate the possibility of the switch is turned off and	
A) disconnected	from the power supply circuit.		
	- ·	g and the center electrode grounded to the engine It lead and waiting the prescribed time.	!

•	•	ected and the center electrode grounded to the the plug and waiting the prescribed time.
249.	L02P	AMP
The constrained g temperature becau		gas turbine engines operates at a cooler
A) it projects into t	he combustion chamber.	
B) the applied volt	age is less.	
C) the construction	n is such that the spark occur	rs beyond the face of the combustion chamber liner.
250.	L02P	AMP
When a magneto i	is operating, what is the prob	able cause for a shift in internal timing
A) The rotating ma	agnet looses its magnetism.	
,	gear teeth are wearing on the	
C) The cam follow	er wear and/or the breaker p	oints wear.
251.	L02P	AMP
If an aircraft ignition probably caused b		e engine continues to run normally, the trouble is
A) an open ground	d lead in the magneto.	
B) arcing magneto	breaker points.	
C) primary lead gr	ounding.	
252.	L03P	AMP
When using an ele	ectric starter motor, the curre	nt flow through it
A) remains relative	ely constant throughout the s	tarting cycle.
B) is highest at the	e start of motor rotation.	
C) is highest just b	pefore starter cutoff (at highes	st RPM.)
253.	L03P	AMP
The purpose of an	under current relay in a star	ter-generator system is to
A) provide a backı	up for the starter relay.	
B) disconnect pow reached.	er from the starter-generator	and ignition when sufficient engine speed is
C) keep current flo	ow to the starter-generator un	der the circuit capacity maximum.
254.	L03P	AMP
(Refer to Powerpla	ant figure 5.) The type of syst	em depicted is capable of operating with
A) external power	only.	

B) either battery o	r external power.	
C) battery power a	and external power simultane	eously.
255.	L03P	AMP
When using an ele	ectric starter motor, current u	sage
A) is highest at the	e start of motor rotation.	
B) remains relative	ely constant throughout the s	tarting cycle.
C) is highest just t	pefore starter cutoff (at higher	st RPM).
256.	L04P	AMP
•	sually employed in pneumation in the sum in	c starters that is used to prevent the starter from nate on schedule is the
A) drive shaft shea	ar point.	
B) stator nozzle de	esign that chokes airflow and	I stabilizes turbine wheel speed.
C) spring coupling	release.	
257.	L04P	AMP
Airflow to the pneo		unit is normally prevented from causing starter
A) stator nozzle do	esign that chokes airflow and	l stabilizes turbine wheel speed.
B) activation of a f	flyweight cutout switch.	
C) a preset timed	cutoff of the airflow at the so	urce.
258.	L04P	AMP
Inspection of pneu	umatic starters by maintenand	ce technicians usually includes checking the
A) oil level and ma	agnetic drain plug condition.	
B) stator and rotor	blades for FOD.	
C) rotor alignment	i.	
259.	L04P	AMP
A clicking sound h		n a pneumatic starter incorporating a sprag clutch
A) gear tooth and/	or pawl damage.	
B) one or more br	oken pawl springs.	
C) the pawls re-co	ontacting and riding on the ra	tchet gear.
260.	L01P	AMP
The purpose of a	safety gap in a magneto is to	
A) prevent burning	g out the primary winding.	

B) protect the high voltage of C) prevent burning of conta		
261.	L01P	AMP
Capacitance afterfiring in m A) fine wire electrodes. B) a built-in resistor in each C) aluminum oxide insulatio		d by the use of
262.	L01P	AMP
As an aircraft engine's spec A) remains constant. B) increases. C) varies with the setting of		ed in the primary coil of the magneto
263.	L01P	AMP
engine, be at its highest val A) Just prior to spark plug f	lue? iring. the spark duration when the flame	, installed on a normally operating e front reaches its maximum velocity.
264.	L01P	AMP
A) Pole shoes, the pole sho B) Primary and secondary o	the magnetic system of a magne- be extensions, and the primary coi coils. le shoes, the pole shoe extensions	l.
265.	L01P	AMP
What is the electrical location A) In parallel with the break B) In series with the breake C) In series with the priman	er points.	gh-tension magneto?
266.	L01P	AMP
What is the radial location of magneto? A) 180° apart.	of the two north poles of a four pole	e rotating magnet in a high tension

B) 270° apart.		
C) 90° apart.		
267.	L01P	AMP
What is the difference	e between a low tension	and a high tension engine ignition system?
A) A low tension syst tension syst	em produces relatively l	low voltage at the spark plug as compared to a high
B) A high tension sys to medium altitude ai		n altitude aircraft, while a low tension system is for low
•		coil near the spark plugs to boost voltage, while the the magneto to the spark plugs.
268.	L01P	AMP
placed in the A) BOTH position.	light to time a magneto	to an aircraft engine, the magneto switch should be
B) OFF position.		
C) LEFT or RIGHT po	osition (either one).	
269.	L01P	AMP
Failure of an engine t	to cease firing after turn	ing the magneto switch off is an indication of
A) an open high tens	ion lead.	
B) an open P-lead to	ground.	
C) a grounded magn	eto switch.	
270.	L01P	AMP
The purpose of stagg	gered ignition is to comp	ensate for
A) short ignition harn	ess.	
B) rich fuel/air mixture	e around exhaust valve.	
C) diluted fuel/air mix	ture around exhaust val	lve.
271.	L01P	AMP
Shielding is used on	spark plug and ignition v	wires to
A) protect the wires f	rom short circuits as a re	esult of chafing or rubbing.
B) prevent outside el	ectromagnetic emission	s from disrupting the operation of the ignition system.
C) prevent interferen	ce with radio reception.	
272.	L01P	AMP
What is the purpose	of a safety gap in some	magnetos?

,	, , , , , , , , , , , , , , , , , , ,	an open occurs in the secondary circuit.
_	e magneto when the ignition swashover in the distributor.	THE TIS OII.
C) TO prevent ha	ishover in the distributor.	
273.	L01P	AMP
A defective prima	ary capacitor in a magneto is ir	ndicated by
A) a fine grained	frosted appearance of the bre	aker points.
B) burned and p	itted breaker points.	
C) a weak spark		
274.	L01P	AMP
What will be the	results of increasing the gap o	f the breaker points in a magneto?
A) Retard the sp	ark and increase its intensity.	
B) Advance the	spark and decrease its intensity	y.
C) Retard the sp	ark and decrease its intensity.	
275.	L01P	AMP
How is the stren	gth of a magneto magnet chec	ked?
•	•	f the primary coil with an ac ammeter while
	agneto at a specified speed.	nainta
	voltage reading at the breaker	
specified speed.	tput of the secondary con with	an ac ammeter while operating the magneto at a
276.	M04P	AMP
What are the poswhen the engine		ve and the dump valve in a jet engine fuel system
A) Pressurization	n valve closed, dump valve ope	∍n.
B) Pressurization	n valve open, dump valve oper	ı.
C) Pressurization	n valve closed, dump valve clo	sed.
277.	M04P	AMP
	r is very important when mixing owing weighs the most?	g fuel and air to obtain a correct fuel to air ratio.
A) 75 parts of dr	y air and 25 parts of water vap	or.
B) 100 parts of c	Iry air.	
C) 50 parts of dr	y air and 50 parts of water vap	or.
278.	M04P	AMP

The economize	r system in a float type carbure	etor
A) keeps the fu	el/air ratio constant.	
B) functions on	ly at cruise and idle speeds.	
C) increases th	e fuel/air ratio at high power se	ettings.
279.	M04P	AMP
In turbine engin	es that utilize a pressurization	and dump valve, the dump portion of the valve
,	flow to the engine fuel manifold the the engine shuts down.	d and dumps the manifold fuel into the combustor to
•	ngine manifold lines to prevent al engine heat (at engine shutd	fuel boiling and subsequent deposits in the lines as a lown).
C) dumps extra throttle advance		provide for quick engine acceleration during rapid
280.	M04P	AMP
A) Decreases e	engine pressure ratio. Compressor and turbine RPM.	ave on the operation of a jet engine?
281.	M04P	AMP
A) Mixture cont B) Automatic m	ise a lean mixture and high cyl rol valve fully closed. ixture control stuck in the exter celerating system.	inder head temperature at sea level or low altitudes? nded position.
282.	M04P	AMP
A) burns too fas	urs when the fuel/air mixture st. e the time of normal ignition.	
283.	M04P	AMP
continuous flow A) RSA system	nce between the Teledyne-Cor fuel injection systems in fuel r uses air pressure only as a m system utilizes airflow as a me	etering force.
•	system uses fuel pressure only	-

284.	M04P	AMP
The primary purpo	ose of the air bleed openings	s used with continuous flow fuel injector nozzles is to
A) provide for auto	omatic mixture control.	
B) lean out the mi	xture.	
C) aid in proper fu	iel vaporization.	
005	MOAD	AMD
285.	M04P	AMP
(Refer to Powerph throughout its ope	•	ost nearly represents an aircraft engine`s fuel/air ratio
A) 1.		
B) 3.		
C) 2.		
286.	M04P	AMP
What method is o	rdinarily used to make idle sp	peed adjustments on a float type carburetor?
	hrottle stop or linkage.	
•	adjustable tapered needle.	
•	·	way which connects the airspace of the float chamber
and the carbureto		·
287.	M04P	AMP
	control is used on most of to	
A) Electromechan		
B) Mechanical.		
C) Hydromechani	cal or electronic.	
288.	M04P	AMP
	sed at rated power in air cool normal cruising range.	led reciprocating engines is richer than the mixture
(2) The mixture us rated power.	sed at idle in air cooled recip	rocating engines is richer than the mixture used at
Regarding the abo	ove statements,	
A) only No. 1 is tru	ue.	
B) only No. 2 is tru	ue.	
C) both No. 1 and	No. 2 are true.	
289.	M04P	AMP
	an normal throttle opening d	uning starting will cause
A) a rich mixture.		

B) a lean mixture.		
C) backfire due to lea	an fuel/air ratio.	
290.	M04P	AMP
The purpose of the b	ack suction mixture con	trol in a float type carburetor is to adjust the mixture by
A) regulating the pre-	ssure drop at the ventur	i.
B) regulating the pre-	ssure on the fuel in the f	loat chamber.
C) regulating the suc	tion on the mixture from	behind the throttle valve.
291.	M04P	AMP
Under which of the fo	ollowing conditions will th	ne trimming of a turbine engine be most accurate?
A) High wind and hig	_J h moisture.	
B) High moisture and	d low wind.	
C) No wind and low i	moisture.	
292.	M02P	AMP
The device that conti	rols the ratio of the fuel/a	air mixture to the cylinders is called a
A) throttle valve.		
B) mixture control.		
C) metering jet.		
293.	M02P	AMP
Select the correct sta	atement concerning the i	dle system of a conventional float type carburetor.
A) The low pressure	area created in the throa	at of the venturi pulls the fuel from the idle passage.
B) Climatic condition	s have very little effect of	n idle mixture requirements.
C) The low pressure the idle passage.	between the edges of the	ne throttle valve and the throttle body pulls the fuel from
294.	M02P	AMP
	•	I if the bellows of the automatic mixture control (AMC) ngine is operating at altitude?
A) It will become lear	ner.	
B) No change will oc	cur until the altitude cha	nges.
C) It will become rich	ier.	
295.	M02P	AMP
The metered fuel pre	essure (chamber C) in ar	n injection type carburetor
A) is held constant th	nroughout the entire eng	ine operating range.

,	ng to the position of the poppe engine driven fuel pump press	et valve located between chamber D (unmetered fuel) ure).
•		n chamber A (impact pressure).
296.	M02P	AMP
What carburetor	component measures the amo	ount of air delivered to the engine?
A) Economizer va	alve.	
B) Automatic mix	ture control.	
C) Venturi.		
297.	M02P	AMP
Fuel is discharge	ed for idling speeds on a float ty	ype carburetor
A) from the idle d	lischarge nozzle.	
B) in the venturi.		
C) through the id	le discharge air bleed.	
298.	M02P	AMP
An aircraft carbui becoming too	retor is equipped with a mixture	e control in order to prevent the mixture from
A) lean at high al	titudes.	
B) rich at high alt	itudes.	
C) rich at high sp	eeds.	
299.	M02P	AMP
Idle cutoff is acco	omplished on a carburetor equi	ipped with a back suction mixture control by
A) introducing lov	w pressure (intake manifold) ai	r into the float chamber.
B) turning the fue	el selector valve to OFF.	
C) the positive cl	osing of a needle and seat.	
300.	M02P	AMP
	float level in a float type carbu	uretor, a measurement is usually made from the top
A) parting surface	e of the carburetor.	
B) top of the float	t.	
C) centerline of the	he main discharge nozzle.	
301.	M02P	AMP
Why must a float	type carburetor supply a rich r	nixture during idle?
A) Engine operat	ion at idle results in higher tha	n normal volumetric efficiency.

B) Because at idline provide proper coo		y not have enough airflow around the cylinders to
C) Because of redu	uced mechanical efficiend	cy during idle.
302.	M02P	AMP
Which of the follow	ring best describes the fu	nction of an altitude mixture control?
A) Regulates the ri	chness of the fuel/air cha	arge entering the engine.
B) Regulates the ai	ir pressure above the fue	el in the float chamber.
C) Regulates the a	ir pressure in the venturi	
303.	M02P	AMP
A reciprocating eng changes in	gine automatic mixture co	ontrol responds to changes in air density caused by
A) altitude or humic	dity.	
B) altitude only.		
C) altitude or tempe	erature.	
304.	M02P	AMP
•	pped with a float type ca cause of the trouble is a	rburetor and the engine runs excessively rich at full clogged
A) main air bleed.		
B) back suction line	€.	
C) atmospheric ver	nt line.	
305.	M02P	AMP
•	e is equipped with a carb ons, the fuel/air mixture	uretor that is not compensated for altitude and will become
A) leaner as either	the altitude or temperatu	re increases.
B) richer as the alti-	tude increases and lean	er as the temperature increases.
C) richer as either t	the altitude or temperatu	re increases.
306.	M02P	AMP
If the main air bleed	d of a float-type carburet	or becomes clogged, the engine will run
A) lean at rated pov	wer.	
B) rich at rated pow	ver.	
C) rich at idling.		
307.	M02P	AMP

What is the possible caus carburetor?	se of an engine running rich a	t full throttle if it is equipped with a float type
A) Float level too low.		
B) Clogged main air bleed	d.	
C) Clogged atmospheric	vent.	
308.	M02P	AMP
Which method is commor	nly used to adjust the level of	a float in a float type carburetor?
A) Lengthening or shorter	ning the float shaft.	
B) Add or remove shims (under the needle valve seat.	
C) Change the angle of the	ne float arm pivot.	
309.	M03P	AMP
Which statement is correct reciprocating engines?	ct regarding a continuous flov	v fuel injection system used on many
A) Fuel is injected directly	into each cylinder.	
B) Fuel is injected at each	n cylinder intake port.	
C) Two injector nozzles a	re used in the injector fuel sy	stem for various speeds.
310.	M03P	AMP
Which of the following car	uses a single diaphragm acco	elerator pump to discharge fuel?
A) An increase in venturi	suction when the throttle valv	re is open.
B) An increase in manifol	d pressure that occurs when	the throttle valve is opened.
C) A decrease in manifold	d pressure that occurs when t	the throttle valve is opened.
311.	M03P	AMP
What is the purpose of the	e carburetor accelerating sys	tem?
A) Supply and regulate th	e fuel required for engine spe	eeds above idle.
B) Temporarily enrich the	mixture when the throttle is	suddenly opened.
C) Supply and regulate a	dditional fuel required for eng	ine speeds above cruising.
312.	M03P	AMP
On a carburetor without a A) be enriched. B) be leaned. C) not be affected.	in automatic mixture control a	as you ascend to altitude, the mixture will
313.	M03P	AMP

What carburetor A) Throttle valve B) Venturi. C) Manifold inta	∂.	desired maximum airflow to the engine at full throttle?
314.	M03P	AMP
What is a function	on of the idling air bleed in a flo	pat type carburetor?
A) It provides a	means for adjusting the mixtur	e at idle speeds.
B) It vaporizes t	he fuel at idling speeds.	
C) It aids in emu	ulsifying/vaporizing the fuel at i	dle speeds.
315.	M03P	AMP
A nine cylinder i which cylinders?		point priming system with a central spider, will prime
A) One, two, thr	ee, eight, and nine.	
B) All cylinders.		
C) One, three, fi	ive, and seven.	
316.	M01P	AMP
A supervisory el information and	lectronic engine control (EEC)	is a system that receives engine operating
A) adjusts a sta operation.	ndard hydromechanical fuel co	ntrol unit to obtain the most effective engine
B) develops the	commands to various actuato	rs to control engine parameters.
C) controls engi	ne operation according to amb	ient temperature, pressure, and humidity.
317.	M01P	AMP
The active clear	rance control (ACC) portion of	an EEC system aids turbine engine efficiency by
A) adjusting stat	tor vane position according to	operating conditions and power requirements.
B) ensuring turb temperatures.	ine blade to engine case clear	ances are kept to a minimum by controlling case
C) automatically	adjusting engine speed to ma	intain a desired EPR.
318.	M01P	AMP
The generally actrimming is to	cceptable way to obtain accura	ate on-site temperature prior to performing engine
A) call the contr	ol tower to obtain field tempera	ature.
B) observe the r	reading on the aircraft Outside	Air Temperature (OAT) gauge.
C) hang a therm	nometer in the shade of the no	se wheel-well until the temperature reading stabilizes.

319.	M01P	AMP
Generally, the p	ractice when trimming an engi	ne is to
A) turn all acces	sory bleed air off.	
B) turn all acces	sory bleed air on.	
•	nents (as necessary) for all en eeither on or off.	igines on the same aircraft with accessory bleed air
320.	N02P	AMP
Kerosene is use	d as turbine engine fuel becaເ	ıse
A) kerosene has	very high volatility which aids	in ignition and lubrication.
B) kerosene has	more heat energy per gallon	and lubricates fuel system components.
•	es not contain any water.	·
321.	N02P	AMP
•	at the fuel pressure fluctuates	and exceeds the upper limits whenever the throttle is e is
A) a ruptured fue	el pump relief valve diaphragm	1.
B) a sticking fuel	l pump relief valve.	
C) an air leak at	the fuel pump relief valve bod	y.
322.	N02P	AMP
		a turbine engine duplex fuel nozzle?
A) Fuel pressure		A
	r the engine reaches idle RPM	1.
C) An electrically	y operated solenoid.	
323.	N02P	AMP
A fuel strainer or	filter must be located betwee	n the
A) boost pump a	and tank outlet.	
B) tank outlet an	d the fuel metering device.	
C) boost pump a	and engine driven fuel pump.	
324.	N02P	AMP
What are the pri	ncipal advantages of the duple	ex fuel nozzle used in many turbine engines?
·		where more efficient and complete burning of the fuel
B) Provides bette	er atomization and uniform flo	w pattern.
•	er range of fuels and filters to b	•

325.	N02P	AMP
What is the purp	oose of the flow divider in a turb	oine engine duplex fuel nozzle?
A) Allows an alto	ernate flow of fuel if the primary	flow clogs or is restricted.
B) Creates the p	orimary and secondary fuel sup	plies.
C) Provides a flo	ow path for bleed air which aids	in the atomization of fuel.
326.	N02P	AMP
What precautior carburetor float		thread lubricant on a tapered pipe plug in a
A) Put the threa	d lubricant only on the first thre	ad.
B) Do not use th	read lubricant on any carburet	or fitting.
C) Engage the f and screw the p		a small amount of lubricant on the second thread
327.	N02P	AMP
Which statemen	it is true regarding proper thrott	le rigging of an airplane?
A) The throttle s	top on the carburetor must be	contacted before the stop in the cockpit.
B) The stop in th	ne cockpit must be contacted b	efore the stop on the carburetor.
C) The throttle of	control is properly adjusted whe	n neither stop makes contact.
328.	N02P	AMP
The fuel system following?	s of aircraft certificated in the s	tandard classification must include which of the
A) An engine dr	iven fuel pump and at least one	auxiliary pump per engine.
B) A positive me	eans of shutting off the fuel to a	Il engines.
•	pply of fuel, available to the endines at least 30 minutes at ME	gine only after selection by the flightcrew, sufficient to FO power.
329.	N02P	AMP
Where physical the fuel line	separation of the fuel lines from	n electrical wiring or conduit is impracticable, locate
A) below the wir	ing and clamp the line securely	to the airframe structure.
B) above the wi	ring and clamp the line securely	to the airframe structure.
C) inboard of the	e wiring and clamp both secure	ly to the airframe structure.
330.	N02P	AMP
What is a chara	cteristic of a centrifugal type fue air and vapor from the fuel.	
, 1	•	

B) It has positive d	isplacement.	
C) It requires a reli	ef valve.	
331.	N01P	AMP
	ne fuel shutoff valve usually	
A) Aft of the firewa	•	
B) Adjacent to the		
C) Downstream of	the engine driven fuel pum	p.
332.	N01P	AMP
Which of the follow supply tank is NOT	•	a centrifugal type fuel boost pump located in a fuel
A) Air and fuel vap	ors do not pass through a	centrifugal type pump.
B) Fuel can be dra	wn through the impeller se	ction of the pump when it is not in operation.
C) The centrifugal	type pump is classified as	a positive displacement pump.
333.	N01P	AMP
(Refer to Powerpla	ant figure 7.) What is the pu	rpose of the fuel transfer ejectors?
A) To supply fuel u	ınder pressure to the engin	e driven pump.
B) To assist in the	transfer of fuel from the ma	ain tank to the boost pump sump.
C) To transfer fuel	from the boost pump sump	to the wing tank.
334.	N01P	AMP
When an electric p	rimer is used, fuel pressure	e is built up by the
A) internal pump ir	the primer solenoid.	
B) suction at the m	ain discharge nozzle.	
C) booster pump.		
335.	N01P	AMP
The primary condit	tion(s) that allow(s) microor	ganisms to grow in the fuel in aircraft fuel tanks is (are)
A) warm temperatu	ures and frequent fueling.	
B) the presence of	water.	
C) the presence of	dirt or other particulate con	ntaminants.
336.	O03P	AMP
If a fire starts in the	e induction system during the	ne engine starting procedure, what should the operator
A) Turn off the fuel	switches to stop the fuel.	

B) Continue crank	ring the engine.	
C) Turn off all swit	tches.	
337.	O03P	AMP
	sing accelerated wear, dust	or sand ingested by a reciprocating engine may also
A) silicon fouling o	of spark plugs.	
B) sludge formation	on.	
C) acid formation.		
338.	O02P	AMP
The purpose of a	bellmouth compressor inlet	is to
A) provide an incr	eased ram air effect at low	airspeeds.
B) maximize the a	erodynamic efficiency of th	e inlet.
C) provide an incr	eased pressure drop in the	inlet.
339.	O02P	AMP
The purpose of a	sonic venturi on a turbocha	rged engine is to
A) limit the amoun	nt of air that can flow from the	ne turbocharger into the cabin for pressurization.
B) increase the an	nount of air that can flow fro	om the turbocharger into the cabin for pressurization.
C) increase the ve	elocity of the fuel/air charge	
340.	O02P	AMP
What is the purpor	se of a turbocharger syster	n for a small reciprocating aircraft engine?
A) Compresses th altitude.	e air to hold the cabin pres	sure constant after the aircraft has reached its critical
B) Maintains cons	tant air velocity in the intak	e manifold.
C) Compresses ai the engine.	ir to maintain manifold pres	sure constant from sea level to the critical altitude of
341.	O02P	AMP
What is the purpor	se of the rate of change co	ntroller in a turbocharger system?
A) Limits the maxi conditions.	mum manifold pressure tha	at can be produced by the turbocharger at full throttle
B) Controls the rate	te at which the turbocharge	er discharge pressure will increase.
C) Controls the po	osition of the waste gate aft	er the aircraft has reached its critical altitude.
342.	O02P	AMP
Boost manifold pro	essure is generally conside	red to be any manifold pressure above

A) 14.7 inches Hg.B) 50 inches Hg.C) 30 inches Hg.		
airplanes that have particle A) Positive and negative ch B) Air/moisture separators,	O02P If to provide clean air to the engines (sand and ice) separators installed arged areas to attract and/or repel pand 'washing' the air clean utilizing change to take advantage of inertial	earticulates out of the airflow. water droplets.
344. What are the three basic re 1. Exhaust bypass assemb 2. Compressor assembly. 3. Pump and bearing casing 4. Density controller. 5. Differential pressure con A) 2, 3, 4. B) 1, 4, 5. C) 1, 2, 3.	g.	AMP boosted turbocharger system?
A) Improper adjustment of 6 B) Excessively rich setting 6 C) Failure of the economize 346. The vortex dissipators insta	carburetor heat valve control linkage on the idle mixture adjustment. er valve to remain closed at takeoff t O02P alled on some turbine-powered aircra	hrottle setting. AMP aft to prevent engine FOD utilize
C) a stream of engine bleed 347. Vortex dissipator systems a A) a landing gear switch.	d air blown toward the ground ahead O02P are generally activated by	d of the engine.

B) a fuel pressure C) an engine inlet	switch anytime an engine airflow sensor.	is operating.
348. What part of an air A) Wing leading ed B) Propeller spinne C) Carburetor.	dge.	AMP accumulate ice before any other?
A) preheat the inta B) mix alcohol with		
350. Increased engine I A) remain the sam B) decrease. C) increase.	P02P heat will cause volumetric ne.	AMP efficiency to
conditions? A) Full open at all B) Full closed at a	times.	AMP engine starting and warmup operations under normal
A) Cylinder head t B) Part throttle ope	P02P ving results in a decrease emperature too low. eration. bes of large diameter.	AMP in volumetric efficiency?
A) A means of coo	P02P on of a blast tube as found bling the engine by utilizing load a cartridge starter.	•

C) A device to cool a	an engine accessory.	
A) excessive cylinde B) increased oil cons	P02P an engine will usually result in er head temperatures. sumption. puildup on spark plugs.	AMP
A) Fully closed. B) Fully open.	P02P ation of an engine, the cowl flaps g to ambient conditions.	AMP should be in what position?
356. A broken cooling fin A) is cause for reject B) may be filed to sn C) should be left alo	tion of the head. nooth contours if damage and/or	AMP repair limits are not exceeded.
A) the downdraft from B) a fan mounted on		AMP d by
358. Which of the following A) Too much cooling B) A cracked cylinder C) Cowling air seal left.	g fin area broken off. er baffle.	AMP ot spot on a reciprocating engine cylinder?
A) a very lean mixtur B) fouled spark plug	P02P emperatures are likely to result f re at high power settings. s. e at high power settings.	AMP

360.	P01P	AMP
How do cowl flaps aid in co	ooling a horizontally opposed aircra	ft engine?
A) Recirculates air through	•	9
B) Directs air through the e	•	
	air flowing around the cylinders.	
361.	P01P	AMP
cooling fin that could be re	uld a mechanic consult to determine moved when cracks are found?	e the maximum amount of cylinder
A) AC 43.13-1A. B) Engine manufacturer's (convice or everboul manual	
	service or overhaul manual.	
C) Engine structure repair	manuai.	
362.	P01P	AMP
Generally, a small crack ju	st started in a cylinder baffle	
A) requires repair by reinfo	orcing, such as installation of a doub	ler over the area.
B) requires no action unles	ss it grows or is branched into two c	racks.
C) may be stop drilled.		
363.	P01P	AMP
The primary purpose of ba engines is to	ffles and deflectors installed around	cylinders of air-cooled aircraft
A) create a low pressure a	rea aft of the cylinders.	
B) force cooling air into clo	se contact with all parts of the cylind	ders.
C) increase the volume of	air used to cool the engine.	
364.	Q02P	AMP
What could be a result of uairplane?	indetected exhaust system leaks in	a reciprocating engine powered
A) Pilot/passenger incapac	citation caused by carbon monoxide	entering the cabin.
B) A rough-running engine	with increased fuel consumption.	
C) Too low exhaust back p	ressure resulting in the desired pov	ver settings not being attained.
365.	Q02P	AMP
	irated engines, turbocharged engine	
A) similar temperatures an		e exhaust systems operate at
B) higher temperatures an	•	
C) similar temperatures an	u pressures.	

366.	Q02P	AMP
How are combustion liner	walls cooled in a gas turbine engine	?
	g through the combustion chamber.	
	and louvers cut in the diffuser sectio	n
C) By bleed air vented fror		•••
o) by blocd all vertica from	in the engine all linet.	
367.	Q02P	AMP
Dislodged internal muffler	baffles on a small reciprocating engi	ine may
A) obstruct the muffler out	let and cause excessive exhaust bac	ck pressure.
B) cause the engine to run	excessively cool.	
C) cause high fuel and oil	•	
,	•	
368.	Q02P	AMP
On an aircraft that utilizes exhaust system be inspect	an exhaust heat exchanger as a souted?	urce of cabin heat, how should the
A) X rayed to detect any c	racks.	
B) Hydrostatically tested.		
C) With the heater air shro	oud removed.	
369.	Q02P	AMP
Power recovery turbines u	sed on some reciprocating engines	are driven by the
A) exhaust gas pressure.		
B) crankshaft.		
C) velocity of the exhaust	gases.	
370.	Q02P	AMP
Select a characteristic of a	good weld on exhaust stacks.	
A) The weld should be bui	lt up 1/8 inch.	
B) Porousness or projectin	ng globules should show in the weld.	
C) The weld should taper of	off smoothly into the base metal.	
371.	Q01P	AMP
Why is high nickel chromic	ım steel used in many exhaust syste	ems?
A) High heat conductivity a	and flexibility.	
B) Corrosion resistance ar	nd low expansion coefficient.	
C) Corrosion resistance ar	nd high heat conductivity.	
372.	Q01P	AMP

	ust system designs commonly used and contraction, may include the use joints.	•
373.	Q01P	AMP
The hot section of a turbine damage? A) Galling. B) Pitting. C) Cracking.	e engine is particularly susceptible to	o which of the following kind of
374. What type of nuts are used A) Brass or heat-resistant r B) High-temperature fiber s C) High-temperature alumin	self-locking nuts.	AMP ylinders?
375.	Q01P	AMP
Sodium filled valves are ad A) are lighter. B) dampen valve impact sh C) dissipate heat well.	vantageous to an aviation engine be	ecause they
376.	Q03P	AMP
A) the engine bleed air sys	r electrical powered compressor.	receive operating pressure from
377.	Q03P	AMP
-	lly true regarding thrust reverser sys	

B) Engine thrust reverse (must all be simultaneou		rcraft usually will not operate independently of each other
C) Mechanical blockage	system design po	ermits a deployment position aft of the exhaust nozzle only.
378.	R03P	AMP
rest in a horizontal positi propeller balancing bear A) Vertical. B) Horizontal.	ion (with the blade	plade propeller to have a persistent tendency to come to es parallel to the ground) while being checked on a
C) Harmonic.		
379.	R03P	AMP
Apparent engine roughn propeller will usually be A) approximately the sau B) greater at low RPM. C) greater at high RPM.	me at all speeds.	ult of propeller unbalance. The effect of an unbalanced
380.	R03P	AMP
Propeller aerodynamic (A) correct blade contour B) static balancing. C) keeping the propeller	ing and angle set	
381.	R02P	AMP
	oropellers reduces der pressure. Thi city characteristics ical stability of gre	s the frictional resistance of moving parts and is easily s statement defines of grease.
382.	R05P	AMP
•	•	onstant speed propeller blades or cuffs must be at least 1/2 and stationary parts of the aircraft. This clearance is with
A) at takeoff pitch (maxii	mum thrust) angle).
B) feathered or in the mo	ost critical pitch co	onfiguration.
C) at the lowest pitch an	gle.	

383.	R05P	AMP
The thrust produced	by a rotating propeller is a	result of
A) an area of low pre	essure behind the propeller	· blades.
B) an area of decrea	sed pressure immediately	in front of the propeller blades.
C) the angle of relative	ve wind and rotational velo	city of the propeller.
384.	R05P	AMP
What is the result of		ciprocating engine when the propeller is in the
A) Opening the throt	tle will cause an increase i	n blade angle.
B) The RPM will vary	directly with any moveme	ent of the throttle.
C) Movement of the	throttle will not affect the b	lade angle.
385.	R05P	AMP
		through the air during one revolution is known as the
A) effective pitch.	p p	
B) geometric pitch.		
C) relative pitch.		
386.	R05P	AMP
• •	on) and which of the follow	te angle between the airfoil section chord line (at the ving?
B) The relative wind.		
	rotation during pitch chang	ne.
o, mo axio oi biado	Totalion daming pilon onang	,
387.	R05P	AMP
How can a steel prop	peller hub be tested for cra	cks?
A) By anodizing.		
B) By magnetic partie	cle inspection.	
C) By etching.		
388.	R05P	AMP
	ng best describes the blade	e movement of a propeller that is in the high RPM
A) Low pitch directly		
,	high pitch to reverse pitch	l.
, .	n feather position to reverse	

389.	R05P	AMP	
Counterweights on constant-speed propellers are generally used to aid in			
A) increasing blade angle.			
B) decreasing blade angle.			
C) unfeathering the propelle	ers.		
390.	R05P	AMP	
	an aircraft using hydromatic full fe		
The feather button, after be opens.	ing pushed, remains depressed un	til the feather cycle is complete, then	
When unfeathering, it is necessary accomplished.	cessary to manually hold the button	down until unfeathering is	
A) Both feather cycle and u	nfeather cycle are functioning prop	erly.	
B) Both feather and unfeath	ner cycles indicate malfunctions.		
C) The feather cycle is corre	ect. The unfeather cycle indicates a	a malfunction.	
204	DOED	AMD	
391. AMP Which of the following best describes the blade movement of a feathering propeller that is in the HIGH RPM position when the feathering action is begun?			
A) High pitch through low pi			
B) Low pitch through revers	se pitch to feather position.		
C) Low pitch through high p	itch to feather position.		
392.	R05P	AMP	
	es or combination of forces operates propeller to the HIGH PITCH position		
A) Engine oil pressure actin acting on the counterweight	ig on the propeller piston cylinder a	rrangement and centrifugal force	
B) Centrifugal force acting of	on the counterweights.		
C) Prop governor oil pressu	ire acting on the propeller piston cy	linder arrangement.	
393.	R05P	AMP	
•	entified as the cambered or curved surface of a wing airfoil section?	side of a propeller blade,	
A) Blade back.			
B) Blade chord.			
C) Blade face.			

394.	R05P	AMP
What controls the constant	speed range of a constant speed p	ropeller?
A) Engine RPM.		
B) Angle of climb and desce	ent with accompanying changes in a	airspeed.
C) The mechanical limits in	the propeller pitch range.	
395.	R05P	AMP
Which of the following defec	cts is cause for rejection of wood pr	opellers?
•	ew heads securing metal tipping.	•
B) An oversize hub or bolth		
C) No protective coating on	-	
396.	R05P	AMP
The primary purpose of a cu		
A) distribute anti icing fluid.	• •	
B) strengthen the propeller.		
C) increase the flow of cool	ing air to the engine nacelle.	
397.	R05P	AMP
	provides maximum efficiency by	7 divil
	the aircraft speed decreases.	
,	most conditions encountered in flig	ıht.
C) increasing the lift coeffici		,
200	R05P	AMP
398. Propoller blade angle is the		AIVIP
Propeller blade angle is the A) chord of the blade and the		
,	ational plane of the propeller.	
	ne rotational plane of the propeller.	
e, energ er and shade and a	retational plane of the proposition	
399.	R05P	AMP
Geometric pitch of a propell	ler is defined as the	
A) effective pitch minus slip	page.	
B) effective pitch plus slippa		
C) angle between the blade	chord and the plane of rotation.	
400.	R05P	AMP
What operational force tend	ls to bend the propeller blades forw	ard at the tip?

A) Torque bending force.		
B) Centrifugal twisting force	ce.	
C) Thrust bending force.		
401.	R05P	AMP
A mechanic certificate and dents on aluminum pr		s the holder to repair deep scars, nicks,
(2) A mechanic certificate straightening of steel prop	with a powerplant rating authorizes beller blades.	s the holder to perform minor
Regarding the above state	ements,	
A) only No. 1 is true.		
B) both No. 1 and No. 2 ar	re true.	
C) neither No. 1 nor No. 2	is true.	
402.	R05P	AMP
The primary purpose of a	feathering propeller is to	
A) prevent further engine	damage when an engine fails in flig	Jht.
B) prevent propeller dama	age when an engine fails in flight.	
	ted by a windmilling propeller wher	n an engine fails in flight.
,	,	
403.	R01P	AMP
Proper operation of electri	ic deicing boots on individual prope	ller blades may best be determined by
A) feeling the boots to see	e if they are heating.	
B) observing the ammeter	or loadmeter for current flow.	
C) feeling the sequence of	f boot heating and have an assistar	nt observe the loadmeter indications.
, .	•	
404.	R01P	AMP
What is a function of the a	automatic propeller synchronizing s	ystem on multiengine aircraft?
A) To control the tip speed	d of all propellers.	
B) To control engine RPM	l and reduce vibration.	
C) To control the power or		
.,	3	
405.	R01P	AMP
On most reciprocating mu through the actuation of the		er synchronization is accomplished
A) throttle levers.		
B) propeller governors.		
C) propeller control levers		
, , , ,		

406.	R01P	AMP
How is aircraft ele propeller hub ass	• • •	eicer systems transferred from the engine to the
A) By slip rings a	nd segment plates.	
B) By slip rings a	nd brushes.	
C) By flexible ele	ctrical connectors.	
407.	R06P	AMP
If a flanged prope	eller shaft has dowel pins	
A) install the prop	peller so that the blades are p	ositioned for hand propping.
B) the propeller c	an be installed in only one po	osition.
C) check carefully	y for front cone bottoming ag	ainst the pins.
400	DOCD	AMD
408.	R06P	AMP
true?	owing statements concerning	the installation of a new fixed pitch wood propeller is
A) If a separate n the propeller.	netal hub is used, final track s	should be accomplished prior to installing the hub in
	erance bolts should be used t	to install the propeller.
C) Inspect the bo	olts for tightness after the first	flight and again after the first 25 hours of flying.
409.	R06P	AMP
Oil leakage arour	nd the rear cone of a hydrom	atic propeller usually indicates a defective
A) piston gasket.		
B) spider shaft oi	l seal.	
C) dome barrel o	il seal.	
410	R06P	AMP
410. Maximum tanar a		
•		nd propeller hub is determined by using
A) bearing blue c	olor transfer.	
B) a micrometer.		
C) a surface gau	ge.	
411.	R06P	AMP
Propeller blade tr	racking is the process of dete	rmining
A) the plane of ro	tation of the propeller with re	spect to the aircraft longitudinal axis.
B) that the blade	angles are within the specific	ed tolerance of each other.
C) the positions o	of the tips of the propeller bla	des relative to each other.

412.	R06P	AMP
-		properly installed and the attachment bolts properly 1/16 inch. The excessive out of track condition may
A) slightly overtig	htening the attachment bolts	adjacent to the most forward blade.
B) discarding the	propeller since out of track c	onditions cannot be corrected.
C) placing shims	between the inner flange and	the propeller.
413.	R06P	AMP
Manually feather	ing a hydromechanical prope	ller means to
A) block governo	r oil pressure to the cylinder o	of the propeller.
B) port governor	oil pressure to the cylinder of	the propeller.
C) port governor	oil pressure from the cylinder	of the propeller.
414.	R06P	AMP
•	essure delivery on a hydroma feathered position?	tic propeller normally stopped after the blades have
A) Pulling out the	feathering push button.	
B) Electric cutout	pressure switch.	
C) Stop lugs in th	ne teeth of the rotating cam.	
415.	R06P	AMP
• .		ly installed hydromatic propeller, it is necessary to control through its entire travel several times to
A) seat the blade	s fully against the low pitch s	top.
B) free the dome	of any entrapped air.	
C) test the maxin	num RPM setting of the gover	rnor.
416.	R07P	AMP
Which of the follo	wing generally renders an al	uminum alloy propeller unrepairable?
A) Any repairs the	at would require shortening a	nd re-contouring of blades.
B) Any slag inclu	sions or cold shuts.	
C) Transverse cr	acks of any size.	
417.	R07P	AMP
One of the advar procedure is that		num propeller utilizing dye-penetrant inspection
A) defects just be	elow the surface are indicated	I.
B) it shows wheth	ner visible lines and other ma	rks are actually cracks rather than scratches.

C) it indicates ov	verspeed condition.	
418.	R07P	AMP
	s, may be repaired by first	area, but not on the leading or trailing edges of
	alf round or flat file.	
	g and applying a proper filler.	
c, caga cama	9 244)9 244	
419.	R07P	AMP
It is important th to	at nicks in aluminum alloy pro	peller blades be repaired as soon as possible in order
A) maintain equa	al aerodynamic characteristics	between the blades.
B) eliminate stre	ess concentration points.	
C) equalize the	centrifugal loads between the	blades.
420.	R07P	AMP
following propell		pellers are not permitted to be made on which of the
A) Shank.		
B) Face.		
C) Back.		
421.	R07P	AMP
Cold straightening	ng a bent aluminum propeller l	blade may be accomplished by
A) the holder of	a mechanic certificate with a p	powerplant rating.
B) an appropriat	ely rated repair station or the i	manufacturer.
C) a person wor and powerplant		the holder of a mechanic certificate with both airframe
422.	R04P	AMP
The propeller go	overnor controls the	
A) oil to and fror	n the pitch changing mechanis	sm.
B) spring tension	n on the boost pump speeder	spring.
C) linkage and c	counterweights from moving in	and out.
423.	R04P	AMP
	fugal force acting on the prope a propeller is in what speed co	eller governor flyweights overcomes the tension on the ondition?

A) On speed.		
B) Underspeed.		
C) Overspeed.		
424.	R04P	AMP
	eed condition of a propeller, the	
	• •	veights is greater than the tension of the speeder
·		ne centrifugal force acting on the governor flyweights is equal to the speeder spring force.
425.	R04P	AMP
can govern in the	eration at speeds lower than INCREASE RPM position, thull HIGH PITCH position.	those for which the constant speed propeller control ne propeller will
,	•	until the HIGH PITCH stop is reached.
,	ull LOW PITCH position.	
426.	R04P	AMP
propeller's consta	nt speed range at a fixed threed threed threed edges are spring) is reduced by r	constant speed propeller is operating within the ottle setting. If the tension of the propeller governor novement of the cockpit propeller control, the
• •		ease, and engine RPM will decrease.
	•	rease, and engine RPM will decrease.
		crease, and engine RPM will increase.
427.	T01P	AMP
	upplied to an APU from	
A) its own indeper	• •	
B) the airplane's r	eserve fuel supply.	
C) the airplane's r	main fuel supply.	
428.	T01P	AMP
Fuel scheduling d maintained	uring APU start and under va	arying pneumatic bleed and electrical loads is
A) manually throu	gh power control lever positi	on.
B) automatically b	y the APU fuel control system	n.
C) automatically b	y an aircraft main engine fue	el control unit.

429.	T01P	AMP

Usually, most of the load placed on an APU occurs when

- A) an electrical load is placed on the generator(s).
- B) the bleed air valve is opened.
- C) the bleed air valve is closed.

430. T01P AMP

When in operation, the speed of an APU

- A) is controlled by a cockpit power lever.
- B) remains at idle and automatically accelerates to rated speed when placed under load.
- C) remains at or near rated speed regardless of the load condition.